

Introduccion A Los Motores Cohete

Propulsion Techniques

Orbital Maneuvers with Finite Thrust Rockets

Experimental Investigation in an Altitude Test Facility of Burning of Excess Combustibles in a Rocket Engine Exhaust

Performance of Two Subliming-solid-propellant Thrustor Systems for Attitude Control of Spacecraft

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Relations of Combustion Dead Time to Engine Variables for a 20,000-pound-thrust Gaseous-hydrogen

A High-performance 250-pound-thrust Rocket Engine Utilizing Coaxial-flow Injection of JP-4 Fuel and Liquid Oxygen

Supersonic Investigation of Nozzle Hinge Moments of a Modified Saturn C-1 Model with and Without Jet Flow

Analysis of Chugging in Liquid-bipropellant Rocket Engines Using Propellants with Different Vaporization Rates

Design and Evaluation of a Turbojet-exhaust Simulator with a Solid-propellant Rocket Motor for Free-flight Research

Investigation of Injectors for a Low-chamber-pressure Hydrogen-fluorine Rocket Engine

17th JANNAF Combustion Meeting, NASA Langley Research Center, Hampton, Virginia, September 22-26, 1980

Máquinas térmicas motoras (volum I)

Design, Development, and Testing of a 1000 Pound (4450 N) Thrust FLOX-propane Ablative Rocket Engine

Evaluation of Tangential Velocity Effects on Spinning Transverse Combustion Instability

Introducción a la ingeniería espacial

Experimental Performance of a Hydrogen-fluorine Rocket Engine at Several Chamber Pressures and Exhaust-nozzle Expansion Area Ratios

Introduccion General a la Tecnologia de Propulsion

Perigee Propulsion for Orbital Launch of Nuclear Rockets

Experimental Performance of Liquid Hydrogen and Liquid Fluorine in Regeneratively Cooled Rocket Engines

Nerva Control Drum Actuator Irradiation Test Program

Introducción a los motores de cohetes

Detonation Control for Propulsion

Solid Propellant Chemistry Combustion and Motor Interior Ballistics 1999

Captive-fired Testing of Solid Rocket Motors

Investigation of Injectors for a Low-chamber- Pressure Hydrogen-fluorine Rocket Engine

Analytical and Experimental Studies of Spherical Solid-propellant Rocket Motors

Comparison of Rocket Performance Using Exhaust Diffuser and Conventional Techniques for Altitude Simulation

Gaseous-hydrogen-liquid-oxygen Rocket Combustion at Supercritical Chamber Pressures

Technology Test Bed

Control of Combustion-chamber Pressure and Oxidant-fuel Ratio for a Regeneratively Cooled Hydrogen-fluorine Rocket Engine

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MILLS CORDOVA

[Propulsion Techniques](#) AIAA

Introducción a la ingeniería espacial proporciona un preámbulo a los fundamentos que constituyen esa especialización de la ingeniería, tanto en el segmento espacial como en el terrestre. Asimismo, provee al lector de las herramientas primordiales para comprender y ampliar los temas relacionados con los sistemas espaciales, desde tópicos elementales como los sistemas de referencias y las unidades temporales utilizados para determinar la posición de los satélites que orbitan la Tierra, hasta disciplinas más sofisticadas como la mecánica celeste y el control térmico de satélites.

Orbital Maneuvers with Finite Thrust Rockets Springer

Drawn from early volumes of *Aerospace America* and its antecedents, this book rescues the insights, concerns, and dreams of dozens of space propulsion experts for the next generation of

aerospace scientists and engineers. Written by well-known figures in space propulsion, this book provides readily accessible source material for design courses in astronautical engineering.

Propulsion Techniques surveys the technologies of rocketry in the traditional categories of liquid, solid, hybrid, nuclear, and electric propulsion. Historical trends and cycles are displayed in each category as articles describe concepts and progress from the early visions of Goddard, Oberth, and Tsiolkovsky to proposed (and re-proposed) ideas for advanced space thrusters. In addition to descriptions of rocket engines of various types, associated technologies for propellants and space-electrical power systems are discussed.

Experimental Investigation in an Altitude Test Facility of Burning of Excess Combustibles in a Rocket Engine Exhaust AIAA

The revised edition of this practical, hands-on book discusses the launch vehicles in use today throughout the world, and includes the latest details on advanced systems being developed, such as electric and nuclear propulsion. The author covers the fundamentals, from the basic principles of rocket propulsion and vehicle dynamics through the theory and practice of liquid and solid

propellant motors, to new and future developments. He provides a serious exposition of the principles and practice of rocket propulsion, from the point of view of the user who is not an engineering specialist.

Performance of Two Subliming-solid-propellant Thrustor Systems for Attitude Control of Spacecraft Univ. Politèc. de Catalunya

This book focuses on the latest developments in detonation engines for aerospace propulsion, with a focus on the rotating detonation engine (RDE). State-of-the-art research contributions are collected from international leading researchers devoted to the pursuit of controllable detonations for practical detonation propulsion. A system-level design of novel detonation engines, performance analysis, and advanced experimental and numerical methods are covered. In addition, the world's first successful sled demonstration of a rocket rotating detonation engine system and innovations in the development of a kilohertz pulse detonation engine (PDE) system are reported. Readers will obtain, in a straightforward manner, an understanding of the RDE & PDE design, operation and testing approaches, and further specific integration schemes for diverse

applications such as rockets for space propulsion and turbojet/ramjet engines for air-breathing propulsion. Detonation Control for Propulsion: Pulse Detonation and Rotating Detonation Engines provides, with its comprehensive coverage from fundamental detonation science to practical research engineering techniques, a wealth of information for scientists in the field of combustion and propulsion. The volume can also serve as a reference text for faculty and graduate students and interested in shock waves, combustion and propulsion.

Comparative In-flight Thrust Measurements of the SERT II Ion Thruster Alpha Editorial Experiments were conducted on an uncooled 20,000lb-thrust gaseous-H and LOX rocket engine over a range of chamber pressure from 45 to 300 psia and oxidant-fuel ratio from 2 to 7. Combustion dead times were measured and compared with dead times calculated from frequency data for two assumed combustion models. Measured combustion dead time decreased with increasing chamber pressure at constant oxidant-fuel ratio or LOX injection velocity. This dead time also decreased with oxidant-fuel ratio at constant chamber pressure or O injection velocity. For the engine model where combustion dead time was considered to be the inverse of twice the

measured chamberpressure frequency, only a fair agreement with the measured dead time was obtained. When the measured chamber-pressure frequencies were corrected for gas-dynamics effects in terms of the gas residence time, close agreement with the measured dead times was obtained. (Author).

Studies of the Exhaust Products from Solid Propellant Rocket Motors Univ. Nacional de Colombia

Calculation of Rocket Vertical-flight Performance Springer Science & Business Media

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